

unit as an integral part of the cabinet assembly.

* * * * *

[FR Doc. 2016-04874 Filed 3-3-16; 8:45 am]

BILLING CODE 6450-01-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2016-4230; Directorate Identifier 2015-CE-041-AD]

RIN 2120-AA64

Airworthiness Directives; EVECTOR, spol. s.r.o. Gliders

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for EVECTOR, spol. s.r.o. Models L 13 SEH VIVAT and L 13 SDM VIVAT gliders (type certificate previously held by AEROTECHNIK s.r.o.) that would supersede AD 2000-20-12. This proposed AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as insufficient material strength of the tail-fuselage attachment fitting. We are issuing this proposed AD to require actions to address the unsafe condition on these products.

DATES: We must receive comments on this proposed AD by April 18, 2016.

ADDRESSES: You may send comments by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.
- *Fax:* (202) 493-2251.
- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.
- *Hand Delivery:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact EVEKTOR, spol. s.r.o., Letecká 1008, 686 04

Kunovice, Czech Republic; phone: +420 572 537 428; email: evektor@evektor.cz; Internet: <http://www.evektor.cz/en/sales-and-support>. You may review copies of the referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329-4148.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2016-4230; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Jim Rutherford, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4165; fax: (816) 329-4090; email: jim.rutherford@faa.gov.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include "Docket No. FAA-2016-4230; Directorate Identifier 2015-CE-041-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD because of those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

On September 28, 2000, we issued AD 2000-20-12, Amendment 39-11923 (65 FR 61262; October 17, 2000) ("AD 2000-20-12"). That AD required actions intended to address an unsafe condition on EVECTOR, spol. s.r.o. Model L 13

SEH VIVAT gliders and was based on mandatory continuing airworthiness information (MCAI) originated by the Civil Aviation Authority, which is the aviation authority for the Czech Republic. That MCAI (AD CAA-AD-T-112/1999R1, dated November 23, 1999), was issued to correct an unsafe condition for EVECTOR, spol. s.r.o. Models L 13 SEH VIVAT and L 13 SDM VIVAT gliders and BLANIK LIMITED Models L-13 Blanik and L-13 AC Blanik gliders. The MCAI states:

To prevent destruction of tail-fuselage attachment fitting which can lead to loss of control of the sailplane. This destruction could be caused due to lower strength of the material used during production.

You may examine the MCAI on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2016-4230.

A review of records since issuance of AD 2000-20-12 revealed that the FAA inadvertently did not address this MCAI for the EVECTOR, spol. s.r.o. Model L 13 SDM VIVAT gliders and the BLANIK LIMITED Model L-13 AC Blanik gliders. This proposed AD would supersede AD 2000-20-12 to add the EVECTOR, spol. s.r.o. Model L 13 SDM VIVAT gliders to the applicability of the AD.

The FAA will address the BLANIK LIMITED Model L-13 AC Blanik gliders in another AD action.

Related Service Information Under 14 CFR Part 51

AEROTECHNIK CZ s.r.o. issued Mandatory Service Bulletin SEH 13-005a, dated November 18, 1999. The service information describes procedures for testing the material strength of attachment fitting part number A 102 021N and instructions for contacting the manufacturer for replacement information if necessary. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section of this NPRM.

FAA's Determination and Requirements of the Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with this State of Design Authority, they have notified us of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all information and determined the unsafe condition exists and is likely to exist or

develop on other products of the same type design.

Costs of Compliance

We estimate that this proposed AD will affect 9 products of U.S. registry. We also estimate that it would take about 4 work-hours per product to comply with the basic requirements of this proposed AD. The average labor rate is \$85 per work-hour. Required parts would cost about \$340 per product.

Based on these figures, we estimate the cost of the proposed AD on U.S. operators to be \$3,060, or \$340 per product.

In addition, we estimate that any necessary follow-on actions would take about 16 work-hours and require parts costing \$500, for a cost of \$1,860 per product. We have no way of determining the number of products that may need these actions.

According to the manufacturer, some of the costs of this proposed AD may be covered under warranty, thereby reducing the cost impact on affected individuals. We do not control warranty coverage for affected individuals. As a result, we have included all costs in our cost estimate.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

(1) Is not a "significant regulatory action" under Executive Order 12866,

(2) Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),

(3) Will not affect intrastate aviation in Alaska, and

(4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by removing Amendment 39–11923 (65 FR 61262; October 17, 2000), and adding the following new AD:

EVECTOR, spol. S.r.o.: Docket No. FAA–2016–4230; Directorate Identifier 2015–CE–041–AD.

(a) Comments Due Date

We must receive comments by April 18, 2016.

(b) Affected ADs

This AD replaces AD 2000–20–12, Amendment 39–11923 (65 FR 61262; October 17, 2000) ("AD 2000–20–12").

(c) Applicability

This AD applies to EVECTOR, spol. s.r.o. Models L 13 SEH VIVAT and L 13 SDM VIVAT gliders (type certificate previously held by AEROTECHNIK s.r.o.), all serial numbers, certificated in any category.

(d) Subject

Air Transport Association of America (ATA) Code 53: Fuselage.

(e) Reason

This AD was prompted by mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as insufficient material strength of the tail-fuselage attachment fitting. We are issuing this proposed AD to detect and correct tail-

fuselage fittings with insufficient material strength, which if left uncorrected could result in detachment of the tail from the fuselage with consequent loss of control.

(f) Actions and Compliance

Unless already done, do the following actions in paragraphs (f)(1) and (f)(2) of this AD, including all subparagraphs:

(1) *Model L 13 SEH VIVAT gliders:*

(i) Within the next 60 days after November 27, 2000 (the effective date retained from AD 2000–20–12), inspect the tail-fuselage attachment fitting, part number (P/N) A 102 021N, for damage and material hardness following the procedures in AEROTECHNIK CZ s.r.o. Mandatory Service Bulletin SEH 13–005a, dated November 18, 1999.

(ii) If you find the tail-fuselage attachment fitting is damaged or the material does not meet the hardness requirements specified in the service bulletin during the inspection required in paragraph (f)(1)(i) of this AD, before further flight, you must contact the manufacturer to obtain an FAA-approved replacement part for P/N A 102 021N and FAA-approved installation instructions and install the replacement part. Use the contact information found in paragraph (h) to contact the manufacturer.

(iii) As of November 27, 2000 (the effective date retained from AD 2000–20–12), do not install, on any glider, a P/N A 102 021N attachment fitting that has not passed the inspection required in paragraph (f)(1)(i) of this AD.

(2) *Model L 13 SDM VIVAT gliders:*

(i) Within the next 60 days after the effective date of this AD, inspect the tail-fuselage attachment fitting, part number (P/N) A 102 021N, for damage and material hardness following the procedures in AEROTECHNIK CZ s.r.o. Mandatory Service Bulletin SEH 13–005a, dated November 18, 1999.

(ii) If you find the tail-fuselage attachment fitting is damaged or the material does not meet the hardness requirements specified in the service bulletin during the inspection required in paragraph (f)(2)(i) of this AD, before further flight, you must contact the manufacturer to obtain an FAA-approved replacement part for P/N A 102 021N and FAA-approved installation instructions and install the replacement part. Use the contact information found in paragraph (h) to contact the manufacturer.

(iii) As of the effective date of this AD, do not install, on any glider, a P/N A 102 021N attachment fitting that has not passed the inspection required in paragraph (f)(2)(i) of this AD.

(g) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, Standards Office, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Jim Rutherford, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4165; fax: (816) 329–4090; email: jim.rutherford@faa.gov. Before

using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) *Airworthy Product*: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(h) Related Information

Refer to MCAI Civil Aviation Authority AD CAA-AD-T-112/1999R1, dated November 23, 1999, for related information. You may examine the MCAI on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2016-4230. For service information related to this AD, contact EVEKTOR, spol. s.r.o., Letecká 1008, 686 04 Kunovice, Czech Republic; phone: +420 572 537 428; email: evektor@evektor.cz; Internet: <http://www.evektor.cz/en/sales-and-support>. You may review copies of the referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329-4148.

Issued in Kansas City, Missouri, on February 24, 2016.

Robert P. Busto,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2016-04543 Filed 3-3-16; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2016-3990; Directorate Identifier 2015-NM-153-AD]

RIN 2120-AA64

Airworthiness Directives; Bombardier, Inc. Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for all Bombardier, Inc. Model CL-600-2D15 (Regional Jet Series 705) and CL-600-2D24 (Regional Jet Series 900) airplanes. This proposed AD was prompted by two in-service incidents reported on Bombardier, Inc. Model CL-600-2C10 (Regional Jet Series 700, 701, & 702) airplanes regarding a loss of all air data information in the flight deck. This proposed AD would require revision of

the airplane flight manual (AFM) to provide procedures to guide the crew to stabilize the airplane's airspeed and attitude for continued safe flight. We are proposing this AD to prevent loss of air data information that may affect continued safe flight.

DATES: We must receive comments on this proposed AD by April 18, 2016.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* 202-493-2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

- *Hand Delivery:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this NPRM, contact Bombardier, Inc., 400 Côte-Vertu Road West, Dorval, Québec H4S 1Y9, Canada; telephone: 514-855-5000; fax: 514-855-7401; email: thd.crj@aero.bombardier.com; Internet: <http://www.bombardier.com>. You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2016-3990; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone: 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Assata Dessaline, Aerospace Engineer, Avionics and Services Branch, ANE-172, FAA, New York Aircraft Certification Office (ACO), 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone: 516-228-7301; fax: 516-794-5531.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include "Docket No. FAA-2016-3990; Directorate Identifier 2015-NM-153-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

Transport Canada Civil Aviation (TCCA), which is the aviation authority for Canada, has issued Canadian AD CF-2015-08, dated April 28, 2015 (referred to after this as the Mandatory Continuing Airworthiness Information, or "the MCAI"), to correct an unsafe condition for all Bombardier, Inc. Model CL-600-2D15 (Regional Jet Series 705) and CL-600-2D24 (Regional Jet Series 900) airplanes. The MCAI states:

Two in-service incidents have been reported on CL-600-2C10 aeroplanes regarding a loss of all air data information in the cockpit. The air data information was recovered as the aeroplane descended to lower altitudes. An investigation determined that the root cause in both events was high altitude icing (ice crystal contamination). If not addressed, this condition may affect continued safe flight.

Due to similarities in the air data systems, such events could happen on all Bombardier CRJ models, CL-600-2B19, CL-600-2C10, CL-600-2D15, CL-600-2D24 and CL-600-2E25. Therefore, the corrective actions for these models will be mandated once their respective Airplane Flight Manual (AFM) revisions become available.

This [Canadian] AD mandates the incorporation of AFM procedures to guide the crew to stabilize the aeroplanes airspeed and attitude for continued safe flight.

Required actions in this NPRM apply only to Bombardier, Inc. Model CL-600-2D15 (Regional Jet Series 705) and CL-600-2D24 (Regional Jet Series 900) airplanes; we may consider issuing further rulemaking on the other Bombardier airplane models identified previously. You may examine the MCAI in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2016-3990.