(e) Reason

This AD was prompted by in-service reports of passenger door tensator spring failures, and qualification testing that determined that incorrect tensator springs could be susceptible to failure prior to reaching their safe-life limit. We are issuing this AD to prevent tensator spring failure, resulting in the inability to open the main passenger door, which could impede evacuation in the event of an emergency.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Maintenance or Inspection Program Revision

Within 30 days after the effective date of this AD, revise the maintenance or inspection program, as applicable, to incorporate the task specified in the Temporary Revisions (TRs) identified in paragraphs (g)(1) through (g)(4) of this AD. The compliance time for doing the initial replacement of the passenger door tensator springs with new springs is at the times specified in the applicable TR specified in paragraphs (g)(1) through (g)(4) of this AD, or within 30 days after the effective date of this AD, whichever occurs later.


(h) No Alternative Actions and Intervals

After accomplishing the revision required by paragraph (g) of this AD, no alternative actions (e.g., inspections) and intervals may be used unless the actions and intervals are approved as an alternative method of compliance (AMOC) in accordance with the procedures specified in paragraph (k)(1) of this AD.

(i) Replacement

For airplanes identified in section 1.A. “Effectivity,” of Bombardier Global 5000 Service Bulletin 700–1A11–52–023, dated October 4, 2013; or Bombardier Global Express/Global Express XRS Service Bulletin 700–52–046, dated October 4, 2013; except as provided by paragraph (j)(1) or (j)(2) of this AD: Within 15 months after the effective date of this AD, but not exceeding the applicable life limit of the passenger tensator spring, replace the passenger door tensator springs having part number (P/N) GS321–0580–1, with new springs, in accordance with the Accomplishment Instructions of Bombardier Global 5000 Service Bulletin 700–1A11–52–023, dated October 4, 2013; or Bombardier Global Express/Global Express XRS Service Bulletin 700–52–046, dated October 4, 2013; as applicable.

(j) Acceptable Alternative Actions for Paragraph (i) of This AD

(1) For airplanes having serial numbers (S/N) 9278 through 9360 inclusive: Replacement of the passenger door tensator springs having P/N GS321–0580–1 with new springs before the effective date of this AD is acceptable for compliance with the requirements of paragraph (i) of this AD. Refer to the task specified in the applicable TRs identified in paragraphs (g)(1) through (g)(4) of this AD for subsequent spring replacements.

For airplanes with serial numbers not identified in paragraph (j)(1) of this AD: Accomplishment after the effective date of this AD of the “Time Limits/Maintenance Checks” discard task identified in the applicable service information specified in paragraphs (g)(1) through (g)(4) of this AD is acceptable for compliance with the requirements of paragraph (i) of this AD.

(k) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) (Alternative Methods of Compliance (AMOCs): The Manager, New York Aircraft Certification Office (ACO), ANE–170, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the New York ACO, send it to ATTN: Program Manager, Continuing Operational Safety, FAA, New York ACO, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone: 516–228–7300; fax: 516–794–5531. Before using any approved AMOC, notify the principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) Contacting the Manufacturer: For any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, New York ACO, ANE–170, FAA; or Transport Canada Civil Aviation (TCCA); or Bombardier, Inc.’s TCCA Design Approval Organization (DAO). If approved by the DAO, the approval must include the DAO-authorized signature.

(l) Related Information


(2) For service information identified in this AD, contact Bombardier, Inc., 400 Côte-Vertu Road West, Dorval, Québec H4S 1Y9, Canada; telephone: 514–855–5000; fax: 514–855–7401; email: ibd.cfr@ aero.bombardier.com; Internet http://www.bombardier.com. You may view this service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221.

Issued in Renton, Washington, on February 19, 2016.

Dorr M. Anderson,
Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2016–04561 Filed 3–3–16; 8:45 am]

BILING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION
Federal Aviation Administration

14 CFR Part 39


RIN 2120–AA64

Airworthiness Directives; Blanik Limited Gliders

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for Blanik Limited Models L–13 Blanik and L–13 AC Blanik gliders (type certificate number (TC) J–16) previously by LET Aeronautical Works) that would supersede AD 99–19–33. This proposed AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as lack of distinct color marking of the elevator drive. We are issuing this proposed AD to require actions to address the unsafe condition on these products.

DATES: We must receive comments on this proposed AD by April 18, 2016.

ADDRESSES: You may send comments by any of the following methods:

• Federal eRulemaking Portal: Go to http://www.regulations.gov. Follow the instructions for submitting comments.

• Fax: (202) 493–2251.


• Hand Delivery: U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590, between 9 a.m.
and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Blanik Limited, 2nd Floor Beauve Lane House, Mercer Street Lower, Dublin 2, Republic of Ireland; phone: +420 733 662 194; email: info@blanik.aero; Internet: http://www.blanik.aero/

You may review this referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329–4148.

Examine the AD Docket

You may examine the AD docket on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA–2016–4233; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647–5527) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Jim Rutherford, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4165; fax: (816) 329–4090; email: jim.rutherford@faa.gov.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the ADDRESSES section. Include “Docket No. FAA–2016–4233; Directorate Identifier 2016–CE–003–AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD because of those comments.

We will post all comments we receive, without change, to http://regulations.gov, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

On November 8, 1999, we issued AD 99–19–33, Amendment 39–11320 (64 FR 50440; September 17, 1999) ("99–19–33"). That AD required actions intended to address an unsafe condition on BLANIK LIMITED Models L–13 Blanik gliders and was based on mandatory continuing airworthiness information (MCAI) originated by the Civil Aviation Authority, which is the aviation authority for the Czech Republic. That MCAI (AD CAA–AD–4–099/98, dated December 30, 1998) was issued to correct an unsafe condition for EVECTOR, spol. s.r.o. Models L 13 SEH VIVAT and L 13 SDM VIVAT gliders and BLANIK LIMITED Models L–13 Blanik and L–13 AC Blanik gliders. The MCAI states:

Colour marking of elevator drive is not inspected or repainted during sailplane operation. The elevator drive is asymmetrical and improper installation causes significant elevator deflection changes.


A review of records since issuance of AD 99–19–33 revealed that the FAA inadvertently did not address this MCAI for the EVECTOR, spol. s.r.o. Models L 13 SEH VIVAT and L 13 SDM VIVAT gliders and the BLANIK LIMITED Model L–13 AC Blanik gliders. This proposed AD would supersede AD 99–19–13 to add the BLANIK LIMITED Model L–13 AC Blanik gliders to the applicability of the AD.

The FAA will address the EVECTOR, spol. s.r.o. Models L 13 SEH VIVAT and L 13 SDM VIVAT gliders in another AD action.

Related Service Information Under 1 CFR Part 51

LET Aeronautical Works has issued LET Mandatory Bulletin MB No.: L13/082a, dated December 10, 1998. The service information describes procedures for painting the left arm of the elevator drive. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the ADDRESSES section of this NPRM.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Costs of Compliance

We estimate that this proposed AD will affect 124 products of U.S. registry. We also estimate that it would take about 1 work-hour per product to comply with the basic requirements of this proposed AD. The average labor rate is $85 per work-hour. Required parts would cost about $10 per product.

Based on these figures, we estimate the cost of the proposed AD on U.S. operators to be $11,780, or $95 per product.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

(1) Is not a “significant regulatory action” under Executive Order 12866;

(2) Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11054, February 26, 1979);

(3) Will not affect intrastate aviation in Alaska, and
(4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39
Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment
Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

§ 39.13 [Amended]

(a) Comments Due Date
We must receive comments by April 18, 2016.

(b) Affected ADs
This AD replaces AD 99–19–33, Amendment 39–11320 (64 FR 50440; September 17, 1999), and adding the following new AD:


(e) Reason
This AD was prompted by mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as lack of distinct color marking of the elevator drive.

(f) Actions and Compliance
Unless already done, do the following actions in paragraphs (f)(1) and (f)(2) of this AD, including all subparagraphs:

(i) Within the next 3 calendar months after November 8, 1999 (the effective date retained from AD 99–19–33), paint the elevator drive mechanism using a contrasting color (such as red) following the procedures in LET Mandatory Bulletin MB No.: L13/082a, dated December 10, 1998.

(ii) As of November 8, 1999 (the effective date retained from AD 99–19–33), only install an elevator bellcrank that has been painted as specified in paragraph (f)(1)(i) of this AD and that has been properly oriented to make sure it is not being installed backward.

(ii) As of the effective date of this AD, only install an elevator bellcrank that has been painted as specified in paragraph (f)(2)(i) of this AD and that has been properly oriented to make sure it is not being installed backward.

(g) Other FAA AD Provisions
The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, Standards Office, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19, Send information to ATTN: Jim Rutherford, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4165; fax: (816) 329–4090; email: jim.rutherford@faa.gov. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) Airworthy Product: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(h) Related Information
Refer to MCAI Civil Aviation Authority AD CAA–AD–4–099/98, dated December 30, 1998, for related information. You may examine the MCAI on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA–2016–4233. For service information related to this AD, contact Blanik Limited, 2nd Floor Beauch Lane House, Mercer Street Lower, Dublin 2, Republic of Ireland; phone: +420 733 662 194; email: info@blanik.aero; Internet: http://www.blanik.aero/%E%/BB%BCustomer_support. You may review this referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329–4148.

Issued in Kansas City, Missouri, on February 24, 2016.

Robert P. Busto,
Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2016–04542 Filed 3–3–16; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[AIRWORTHINESS DIRECTIVES; EVEKTOR, spol. s.r.o. Gliders]

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for EVEKTOR, spol. s.r.o. Model L 13 SEH VIVAT and L 13 SDM VIVAT gliders (type certificate previously held by AEROTECHNIK s.r.o.). This proposed AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as lack of distinct color marking of the elevator drive. We are issuing this proposed AD to require actions to address the unsafe condition on these products.

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Fax: (202) 493–2251.


Hand Delivery: U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact EVEKTOR, spol. s.r.o., Letecka 1008, 686 04 Kunovice, Czech Republic; phone: +420