

Issued in Renton, Washington, on May 3, 2016.

Michael Kaszycki,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2016-6431; Directorate Identifier 2015-NM-182-AD]

RIN 2120-AA64

Airworthiness Directives; Airbus Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for certain Airbus Model A318-112 airplanes, A319-111, -112, -115, -132, and -133 airplanes, A320-214, -232, and -233 airplanes, and A321-211, -212, -213, -231, and -232 airplanes. This proposed AD was prompted by a quality control review on the final assembly line, which determined that aluminum alloy with inadequate heat treatment had been delivered and used on several structural parts. This proposed AD would require a one-time eddy current conductivity measurement of certain cabin, cargo compartment, and frame structural parts to determine if aluminum alloy with inadequate heat treatment was used, and replacement if necessary. We are proposing this AD to detect and replace structural parts made of aluminum alloy with inadequate heat treatment. This condition could result in reduced structural integrity of the airplane.

DATES: We must receive comments on this proposed AD by June 27, 2016.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- **Federal eRulemaking Portal:** Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- **Fax:** 202-493-2251.

- **Mail:** U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

- **Hand Delivery:** Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this NPRM, contact Airbus, Airworthiness Office—EIAS, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone: +33 5 61 93 36 96; fax: +33 5 61 93 44 51; email: account.airworth-eas@airbus.com; Internet: <http://www.airbus.com>. You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2016-6431; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone: 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT:

Sanjay Ralhan, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone: 425-227-1405; fax: 425-227-1149.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include "Docket No. FAA-2016-6431; Directorate Identifier 2015-NM-182-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Union, has issued EASA Airworthiness

Directive 2015-0219, dated November 3, 2015 (referred to after this as the Mandatory Continuing Airworthiness Information, or "the MCAI"), to correct an unsafe condition for certain Airbus Model A318-112 airplanes, A319-111, -112, -115, -132, and -133 airplanes, A320-214, -232, and -233 airplanes, and A321-211, -212, -213, -231, and -232 airplanes. The MCAI states:

Following an Airbus quality control review on the final assembly line, it was discovered that aluminum alloy with inadequate heat treatment were delivered by a supplier for several structural parts. The results of the investigations highlighted that 1% of the stock could be impacted by this wrong material.

Structural investigations demonstrated the capability to sustain the static limits loads, and sufficient fatigue life up to a certain inspection threshold.

This condition, if not detected and corrected, could reduce the aeroplane structural integrity following fatigue load.

To address this potential unsafe condition, Airbus issued Service Bulletin (SB) A320-53-1292, SB A320-53-1293, and SB A320-53-1294 to provide inspection instructions.

For the reasons described above, this [EASA] AD requires a one-time Special Detailed Inspection (SDI) [*i.e.*, eddy current conductivity measurement] of certain cabin, cargo compartment and frame parts [for material identification] and, depending on findings, replacement with serviceable parts.

You may examine the MCAI in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2016-6431.

Related Service Information Under 14 CFR Part 51

Airbus has issued the following service information:

- Airbus Service Bulletin A320-53-1292, dated July 23, 2015; including Appendices 01 and 02, dated July 23, 2015.

- Airbus Service Bulletin A320-53-1293, dated July 30, 2015; including Appendices 01 and 02, dated July 30, 2015.

- Airbus Service Bulletin A320-53-1294, dated July 23, 2015; including Appendices 01 and 02, dated July 23, 2015.

The service information describes procedures for a one-time eddy current conductivity measurement of certain cabin, cargo compartment, and frame structural parts to determine if aluminum alloy with inadequate heat treatment was used, and replacement of any affected part with a serviceable part. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

FAA’s Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all pertinent information and determined an unsafe condition exists and is likely to exist or develop on other products of these same type designs.

Costs of Compliance

We estimate that this proposed AD affects 46 airplanes of U.S. registry.

We also estimate that it would take about 6 work-hours per product to comply with the basic requirements of this proposed AD. The average labor rate is \$85 per work-hour. Based on these figures, we estimate the cost of this proposed AD on U.S. operators to be \$23,460, or \$510 per product.

We have received no definitive data that would enable us to provide cost estimates for the on-condition actions specified in this proposed AD.

According to the manufacturer, some of the costs of this proposed AD may be covered under warranty, thereby reducing the cost impact on affected individuals. We do not control warranty coverage for affected individuals. As a result, we have included all available costs in our cost estimate.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority

because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

1. Is not a “significant regulatory action” under Executive Order 12866;
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska; and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

Airbus: Docket No. FAA–2016–6431; Directorate Identifier 2015–NM–182–AD.

(a) Comments Due Date

We must receive comments by June 27, 2016.

(b) Affected ADs

None.

(c) Applicability

This AD applies to the Airbus airplanes identified in paragraphs (c)(1) through (c)(4) of this AD, certificated in any category; manufacturer serial numbers 4895, 4903, 4911, 4919, 4929, 4938, 4942, 4944, 4946, 4948, and 4951, 4956 through 5541 inclusive, 5544, 5547, 5550, 5551, 5553, 5556, 5559, 5561, 5562, 5563, 5565, 5566, 5570, 5572, 5576, and 5578.

- (1) Airbus Model A318–112 airplanes.
- (2) Airbus Model A319–111, –112, –115, –132, and –133 airplanes.
- (3) Airbus Model A320–214, –232, and –233 airplanes.
- (4) Airbus Model A321–211, –212, –213, –231, and –232 airplanes.

(d) Subject

Air Transport Association (ATA) of America Code 53, Fuselage.

(e) Reason

This AD was prompted by a quality control review on the final assembly line, which determined that aluminum alloy with inadequate heat treatment had been delivered and used on several structural parts. We are issuing this AD to detect and replace structural parts made of aluminum alloy with inadequate heat treatment. This condition could result in reduced structural integrity of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) One-time Measurement

Within 6 years since the date of issuance of the original certificate of airworthiness or the date of issuance of the original export certificate of airworthiness: Do a one-time eddy current conductivity measurement of the cabin, cargo compartment, and frame structural parts identified in the “Affected P/N (part number)” column of tables 1, 2, and 3 to paragraphs (g) and (h) of this AD to determine if aluminum alloy with inadequate heat treatment was used, in accordance with the Accomplishment Instructions of the applicable service information identified in paragraphs (g)(1), (g)(2), and (g)(3) of this AD.

- (1) For cabin structural parts: Airbus Service Bulletin A320–53–1292, dated July 23, 2015; including Appendices 01 and 02, dated July 23, 2015.
- (2) For cargo compartment structural parts: Airbus Service Bulletin A320–53–1293, dated July 30, 2015; including Appendices 01 and 02, dated July 30, 2015.
- (3) For frame structural parts: Airbus Service Bulletin A320–53–1294, dated July 23, 2015; including Appendices 01 and 02, dated July 23, 2015.

TABLE 1 TO PARAGRAPHS (G) AND (H) OF THIS AD—PARTS TO BE INSPECTED/INSTALLED
[Airbus Service Bulletin A320–53–1292]

Affected P/N	Acceptable replacement P/N	Area
D212724550000	D212724550000	Cabin
D2127247600200	D2127247600200	Cabin

TABLE 1 TO PARAGRAPHS (G) AND (H) OF THIS AD—PARTS TO BE INSPECTED/INSTALLED—Continued
 [Airbus Service Bulletin A320–53–1292]

Affected P/N	Acceptable replacement P/N	Area
D2127247600300	D2127247600300	Cabin
D2127399900200	D2127399900200	Cabin
D2127399900300	D2127399900300	Cabin
D2127698900800	D2127698900800	Cabin
D2127698902400	D2127698902400	Cabin
D2527075131200	D2527075131251	Cabin
D2527075131300	D2527075131351	Cabin
D2527075138000	D2527075138000	Cabin
D2527075138100	D2527075138100	Cabin
D2527075138200	D2527075138200	Cabin
D2527075138300	D2527075138300	Cabin
D2527075138600	D2527075138651	Cabin
D2527075138800	D2527075138851	Cabin
D2527240220600	D2527240220651	Cabin
D2527240220700	D2527240220751	Cabin
D2527240220800	D2527240220851	Cabin
D9249591201000	D9249591201000	Cabin
D9249591201800	D9249591201800	Cabin
D9249591227800	D9249591227851	Cabin
D9249591227900	D9249591227951	Cabin
D9249591228000	D9249591228051	Cabin
D9249591228100	D9249591228151	Cabin

TABLE 2 TO PARAGRAPHS (G) AND (H) OF THIS AD—PARTS TO BE INSPECTED/INSTALLED
 [Airbus Service Bulletin A320–53–1293]

Affected P/N	Acceptable replacement P/N	Area
D2707033520000	D2707033520000	Cargo
D2827027120000	D2827027120000	Cargo
D2827093500400	D2827093500400	Cargo
D2907013701200	D2907013701251	Cargo
D2907013800400	D2907013800451	Cargo
D3247012900000	D3247012900051	Cargo
D3817003820000	D3817003820000	Cargo
D3817012320200	D3817012320251	Cargo
D3837021201600	D3837021201600	Cargo
D3837033300400	D3837033300400	Cargo
D4918518320200	D4918518320200	Cargo
D5347043420400	D5347043420451	Cargo
D9248511000000	D9248511000051	Cargo
D9249254100200	D9249254100251	Cargo
D9249282300000	D9249282300000	Cargo

TABLE 3 TO PARAGRAPHS (G) AND (H) OF THIS AD—PARTS TO BE INSPECTED/INSTALLED
 [Airbus Service Bulletin A320–53–1294]

Affected P/N	Acceptable replacement P/N	Area
D2827098326800	D2827098326851	Frame
D5347051620600	D5347051620651	Frame
D5347051720600	D5347051720651	Frame
D5347057120000	D5347057120051	Frame
D5347067520600	D5347067520651	Frame
D5347067521400	D5347067521451	Frame
D5347067520800	D5347067520851	Frame
D5347067521000	D5347067521051	Frame
D5347067521600	D5347067521651	Frame
D5347067620600	D5347067620600	Frame
D5347067720200	D5347067720251	Frame
D5347067720400	D5347067720451	Frame
D5347986520200	D5347986520251	Frame

(h) Replacement

If during the measurement required by paragraph (g) of this AD, any affected P/N specified in table 1, 2, or 3 to paragraphs (g) and (h) of this AD is found to have a measured value greater than that specified in Figure A–GFAAA, Sheet 01, “Inspection Flowchart,” of the applicable service information identified in paragraphs (g)(1), (g)(2), and (g)(3) of this AD: Before further flight, replace the affected part with the corresponding acceptable replacement part specified in table 1, 2, or 3 to paragraphs (g) and (h) of this AD, in accordance with the Accomplishment Instructions of the applicable service information identified in paragraphs (g)(1), (g)(2), and (g)(3) of this AD.

(i) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) **Alternative Methods of Compliance (AMOCs):** The Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to ATTN: Sanjay Ralhan, Aerospace Engineer, International Branch, ANM–116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057–3356; telephone: 425–227–1405; fax: 425–227–1149. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) **Contacting the Manufacturer:** For any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA; or the EASA; or Airbus’s EASA DOA. If approved by the DOA, the approval must include the DOA-authorized signature.

(3) **Required for Compliance (RC):** If any service information contains procedures or tests that are identified as RC, those procedures and tests must be done to comply with this AD; any procedures or tests that are not identified as RC are recommended. Those procedures and tests that are not identified as RC may be deviated from using accepted methods in accordance with the operator’s maintenance or inspection program without obtaining approval of an AMOC, provided the procedures and tests identified as RC can be done and the airplane can be put back in an airworthy condition. Any substitutions or changes to procedures or tests identified as RC require approval of an AMOC.

(j) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) EASA AD 2015–0219, dated November 3, 2015, for

related information. This MCAI may be found in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2016–6431.

(2) For service information identified in this AD, contact Airbus, Airworthiness Office—ELAS, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone: +33 5 61 93 36 96; fax: +33 5 61 93 44 51; email: account.airworth-eas@airbus.com; Internet: <http://www.airbus.com>. You may view this service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221.

Issued in Renton, Washington, on May 4, 2016.

Michael Kaszycki,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

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SOCIAL SECURITY ADMINISTRATION**20 CFR Parts 404, 411, and 416**

[Docket No. SSA–2014–0016]

RIN 0960–AH66

Unsuccessful Work Attempts and Expedited Reinstatement Eligibility

AGENCY: Social Security Administration.
ACTION: Notice of proposed rulemaking.

SUMMARY: We propose to remove some of the requirements for evaluation of an unsuccessful work attempt (UWA) that lasts between 3 and 6 months. We also propose to allow previously entitled beneficiaries to apply for expedited reinstatement (EXR) in the same month they stop performing substantial gainful activity (SGA). Provisional benefits will begin the month after the request for EXR if the beneficiary stops performing SGA in the month of the EXR request. These changes would simplify our policies and make them easier for the public to understand.

DATES: To ensure that your comments are considered, we must receive them no later than July 11, 2016.

ADDRESSES: You may submit comments by any one of three methods—Internet, fax, or mail. Do not submit the same comments multiple times or by more than one method. Regardless of which method you choose, please state that your comments refer to Docket No. SSA–2014–0016 so that we may associate your comments with the correct regulation.

CAUTION: You should be careful to include in your comments only information that you wish to make publicly available. We strongly urge you

not to include in your comments any personal information, such as Social Security numbers or medical information.

1. **Internet:** We strongly recommend that you submit your comments via the Internet. Please visit the Federal eRulemaking portal at <http://www.regulations.gov>. Use the Search function to find docket number SSA–2014–0016. The system will issue a tracking number to confirm your submission. You will not be able to view your comment immediately because we must post each comment manually. It may take up to a week for your comment to be viewable.

2. **Fax:** Fax comments to (410) 966–2830.

3. **Mail:** Mail your comments to the Office of Regulations and Reports Clearance, Social Security Administration, 3100 West High Rise Building, 6401 Security Boulevard, Baltimore, Maryland 21235–6401.

Comments are available for public viewing on the Federal eRulemaking portal at <http://www.regulations.gov> or in person, during regular business hours, by arranging with the contact person identified below.

FOR FURTHER INFORMATION CONTACT: Kristine Erwin-Tribbitt, Office of Retirement and Disability Policy, Office of Research, Demonstration, and Employment Support, Social Security Administration, 6401 Security Boulevard, Robert Ball Building 3–A–26, Baltimore, MD 21235–6401, (410) 965–3353. For information on eligibility or filing for benefits, call our national toll-free number, 1–800–772–1213 or TTY 1–800–325–0778, or visit our Internet site, Social Security Online, at <http://www.socialsecurity.gov>.

SUPPLEMENTARY INFORMATION:**SGA and UWA**

To be eligible for disability benefits, an individual must be unable to engage in any SGA.¹ SGA is work activity that is both substantial and gainful.² Work activity is substantial if it involves the performance of significant physical or mental activities.³ “Gainful work activity” is work done for pay or profit, or if it is the kind of work usually done for pay or profit, whether or not a profit is realized.⁴ We will not determine that an individual is disabled or continues to be disabled if he or she is able to perform SGA.

We use several rules to decide whether an individual has performed

¹ 42 U.S.C. 223(d)(1), 42 U.S.C. 1382c(a)(3)(A).

² 20 CFR 404.1572 and 416.972.

³ 20 CFR 404.1572(a) and 416.972(a).

⁴ 20 CFR 404.1572(b) and 416.972(b).