paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.
(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.
(3) The following service information was approved for IBR on August 12, 2016.
(4) The following service information was approved for IBR on October 22, 2012 (77 FR 57003, September 17, 2012).
(ii). Reserved.
(5) For Airbus service information identified in this AD, contact Airbus, Airworthiness Office—EIAS, 1 Rond Point Maurice Bellonte, 31070 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; email account.airworth-eas@airbus.com; Internet http://www.airbus.com.
(7) You may view this service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1227.
(8) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202–741–6030, or go to: http://www.archives.gov/federal-register/cfr/ibr-locations.html.

 Issued in Renton, Washington, on June 23, 2016.

Dorr M. Anderson,
Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2016–15902 Filed 7–7–16; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION
Federal Aviation Administration

14 CFR Part 39


RIN 2120–AA64

Airworthiness Directives; Pacific Aerospace Limited Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: We are superseding Airworthiness Directive (AD) 2006–13–05 for certain Pacific Aerospace Limited Model 750XL (type certificate previously held by Pacific Aerospace Corporation Ltd.) airplanes. This AD results from mandatory continuing airworthiness information (MCAI) issued by an aviation authority of another country to identify and correct an unsafe condition on an airplane product. The MCAI describes the unsafe condition as some critical rivets on the wing not being fully age-hardened and being installed in specific locations where reduction in rivet strength reduces wing strength. We are issuing this AD to require actions to address the unsafe condition on these products.

DATES: This AD is effective August 12, 2016.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of August 12, 2016.

The Director of the Federal Register approved the incorporation by reference of a certain other publication listed in this AD as of July 31, 2006 (71 FR 35509, June 21, 2006).


For service information identified in this AD, contact Pacific Aerospace Limited, Airport Road, Hamilton, Private Bag 3027, Hamilton 3240, New Zealand; telephone: +64 7 843 6144; facsimile: +64 7 843 6134; email: pacific@aerospace.co.nz; Internet: www.aerospace.co.nz. You may view this referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329–4148. It is also available on the Internet at http://www.regulations.gov by searching for Docket No. FAA–2016–5578.

FOR FURTHER INFORMATION CONTACT: Karl Schletzbaum, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4123; fax: (816) 329–4090; email: karl.schletzbaum@faa.gov.

SUPPLEMENTARY INFORMATION:

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to certain Pacific Aerospace Limited Model 750XL (type certificate previously held by Pacific Aerospace Corporation Ltd.) airplanes. That NPRM was published in the Federal Register on April 12, 2016 (81 FR 21489), and proposed to supersede AD 2006–13–05, Amendment 39–14658 (71 FR 35509, June 21, 2006) (“AD 2006–13–05”).

Since we issued AD 2006–13–05, additional airplanes have been identified that need to be added to the applicability of the AD.

The Civil Aviation Authority (CAA), which is the aviation authority for New Zealand, has issued AD No. DCA/750XL/7B, dated February 25, 2016 (referred to after this as “the MCAI”), to correct an unsafe condition for the specified products. The MCAI states:

DCA/750XL/7B revised to introduce PACSB/XL/018 issue 4, dated 20 January 2016, which reduces the applicability to S/N 101 through to 131 with no change to the requirements. Aircraft with S/N 132 onwards have been modified in accordance with PACSB/XL/018 at manufacture, which is a terminating action for the requirements of this AD.

This AD requires you to remove rivets that have not been fully age hardened and replace them with bolts, washers, and nuts in specific locations where reduction in rivet strength affects overall structural capability. The AD retains the airplane weight AFM limitations until the rivets are replaced with the bolts, washers, and nuts. You may examine the MCAI on the Internet at https://www.regulations.gov/#docketDetail;D=FAA-2016-5578-002.

Comments

We gave the public the opportunity to participate in developing this AD. We received no comments on the NPRM (81 FR 21489, April 12, 2016) or on the determination of the cost to the public.

Conclusion

We reviewed the relevant data and determined that air safety and the public interest require adopting the AD as proposed except for minor editorial changes. We have determined that these minor changes:

• Are consistent with the intent that was proposed in the NPRM (81 FR 21489, April 12, 2016) for correcting the unsafe condition; and

Do not add any additional burden upon the public than was already proposed in the NPRM (81 FR 21489, April 12, 2016).
Related Service Information

We reviewed Pacific Aerospace Limited Service Bulletin PACSB/XL/018, Issue 4, dated January 20, 2016. The service bulletin describes procedures for removing rivets (part number (P/N) MS20470 DD6) and installing bolts (P/N NAS 6203–7X or NAS 6203–6X), washers (P/N AN960–10), and nuts (P/N MS21044N3) in place of the rivets to restore airplane to full take-off weight of 7,500 pounds. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the ADDRESSES section of the AD.

Costs of Compliance

We estimate that this AD will affect 9 products of U.S. registry. We also estimate that it will take about 32 work-hours per product to comply with the replacement requirements of this AD. The average labor rate is $85 per work-hour. Required parts will cost about $519 per product.

Based on these figures, we estimate the cost of this AD on U.S. operators to be $29,151, or $3,239 per product.

AD 2006–13–05 affected 8 of the 9 U.S.-registered airplanes reflected in the above cost information. This AD will only increase the cost already required by AD 2006–13–05 by one additional airplane. The FAA has a report that the additional airplane is already in compliance, thus this AD will impose no additional cost impact on U.S. operators.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this AD:

(1) Is not a “significant regulatory action” under Executive Order 12866,

(2) Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),

(3) Will not affect intrastate aviation in Alaska,

(4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

Examining the AD Docket

You may examine the AD docket on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA–2016–5578; in person at the Docket Office (telephone (800) 647–5578; or in person at the Docket Office (telephone (800) 647–5578) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

§ 39.13 [Amended]

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by removing Amendment 39–14658 (71 FR 35509, June 21, 2006), and adding the following new AD:


(a) Effective Date

This airworthiness directive (AD) becomes effective August 12, 2016.

(b) Affected ADs


(c) Applicability

This AD applies to the following Pacific Aerospace Limited Model 750XL airplanes (type certificate previously held by Pacific Aerospace Corporation, that are certificated in any category. (1) Airplanes previously affected by AD 2006–13–05: Serial numbers 101, 102, 104 through 120, and 125.

(2) Airplanes new to this AD: Serial numbers 103, 121, 122, 123, 124, and 126 to 131.

(d) Subject

Air Transport Association of America (ATA) Code 57: Wings.

(e) Reason

This AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as some critical rivets on the wing not being fully age-hardened and being installed in specific locations where reduction in rivet strength reduces wing strength. We are issuing this AD to add airplane serial numbers to the Applicability section, paragraph (c) of this AD, and to ensure wing ultimate load requirements are met. If wing ultimate load requirements are not met, wing failure could result with consequent loss of control.

(f) Actions and Compliance

Unless already done, do the following actions:

(1) Insert the following information into the Limitations section of the airplane flight manual (AFM) at the compliance time specified in paragraphs (i)(1)(i) and (ii) of this AD. You may do this by inserting a copy of this AD into the Limitations section of the AFM: “The maximum takeoff weight is reduced from 7,500 pounds to 7,125 pounds.”

The owner/operator holding at least a private pilot certificate as authorized by section 43.7 of the Federal Aviation Regulations (14 CFR 43.7) may do the flight manual changes requirement of this AD. Make an entry in the aircraft records showing compliance with this portion of the AD following section 43.9 of the Federal Aviation Regulations (14 CFR 43.9).

(i) For airplanes previously affected by AD 2006–13–05: Before further flight after January 16, 2006 (the effective date retained from AD 2005–26–53, Amendment 39–14451 (71 FR 2453, January 17, 2006), which was replaced by AD 2006–13–05).

(ii) For airplanes new to this AD: Before further flight after August 12, 2016 (the effective date of this AD).

(2) Remove rivets, part number (P/N) MS20470 DD6, on the main spar web and

(3) Remove rivets, part number (P/N) MS20470 DD6, on the main spar web and

(4) Remove rivets, part number (P/N) MS20470 DD6, on the main spar web and
replace with bolts, P/N NAS 6203–6X or —7X, as indicated for the position, assembled with washers, P/N AN960–10, and nut, P/N MS21044N3, at the compliance time specified in paragraphs (f)(2)(i) and (ii) of this AD.


(ii) For airplanes new to this AD: Within the next 100 hours TIS after August 12, 2016 (the effective date of this AD) or within the next 12 months after August 12, 2016 (the effective date of this AD), whichever occurs first. Do the removal and replacement actions following Pacific Aerospace Limited Service Bulletin PACSB/XL/018, Issue 4, dated January 20, 2016.

(3) For all affected airplanes: Before further flight after doing the action required in paragraph (f)(2)(i) of this AD, remove the restrictive information from the Limitations section of the AFM that you were required to insert in paragraph (f)(1) of this AD. The owner/operator holding at least a private pilot certificate as authorized by section 43.7 of the Federal Aviation Regulations (14 CFR 43.7) may do the flight manual changes requirement of this AD.

(g) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, Standards Office, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Karl Schletzbaum, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4123; facsimile: (816) 329–4090; email: karl.schletzbaum@faa.gov. Before using any approved AMOC on any aircraft to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) Airworthiness Product: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use those actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(h) Related Information

Refer to MCAI Civil Aviation Authority (CAA) AD No. DCA/750XL/7B, dated February 25, 2016, for related information. You may examine the MCAI on the Internet at https://www.regulations.gov/#!docketDetail;D=FAA-2016-5578-002.

(i) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51. (2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(3) The following service information was approved for RR on August 12, 2016.


(ii) Reserved.

(4) The following service information was approved for RR on August 12, 2006 (71 FR 35509, June 21, 2006).


(ii) Reserved.

(5) For Pacific Aerospace Limited service information identified in this AD, contact Pacific Aerospace Limited, Airport Road, Hamilton, Private Bag 3027, Hamilton 3240, New Zealand; telephone: +64 7 843 6144; facsimile: +64 7 843 6134; email: pacific@aerospace.co.nz; Internet: www.aerospace.co.nz.

(6) You may view this service information at FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329–4148. In addition, you can access this service information on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA–2016–5578.

(7) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202–741–6030, or go to: http://www.archives.gov/federal-register/cfr/ibr-locations.html.

Issued in Kansas City, Missouri, on June 28, 2016.

Pat Mullen,
Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2016–15864 Filed 7–7–16; 8:45 am]
BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39


RIN 2120–AA64

Airworthiness Directives; Beechcraft Corporation (Type Certificate Previously Held by Hawker Beechcraft Corporation; Raytheon Aircraft Company) Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for certain Beechcraft Corporation Model BAe.125 Series 1000A and 1000B airplanes and Model Hawker 1000 airplanes. This AD was prompted by reports of inadvertent stowage of the thrust reversers, which can result in high forward engine thrust even though the throttle is commanding reverse thrust. This AD requires installing kits that include relays, associated wiring, and a thrust reverser fail annunciator. We are issuing this AD to prevent inadvertent stowage of the thrust reversers, which could cause a runway overrun during a rejected takeoff or landing, and consequent structural failure and possible injury to occupants.

DATES: This AD is effective August 12, 2016.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of August 12, 2016.

ADDRESSES: For service information identified in this final rule, contact Beechcraft Corporation, TMDC, P.O. Box 85, Wichita, KS 67201–0085; telephone: 316–676–8238; fax: 316–671–2540; email: tmdc@beechcraft.com; Internet: http://pubs.beechcraft.com. You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221. It is also available on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA–2016–0460.

Examining the AD Docket
You may examine the AD docket on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA–2016–0460; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (phone: 800–647–5527) is Docket Management Facility, U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SW., Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT: Jeffrey Englert, Aerospace Engineer, Systems and Propulsion Branch, ACE–116W, FAA, Wichita Aircraft Certification Office (ACO), 1801 Airport Road, Room 100, Dwight D. Eisenhower