products identified in this rulemaking action.

**Regulatory Findings**

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

1. Is not a “significant regulatory action” under Executive Order 12866,
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),
3. Will not affect intrastate aviation in Alaska, and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**The Proposed Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

1. The authority citation for part 39 continues to read as follows:
   
   Authority: 49 U.S.C. 106(g), 40113, 44701.

   **§39.13 [Amended]**

2. The FAA amends §39.13 by adding the following new airworthiness directive (AD):


- **(a) Comments Due Date**

  We must receive comments by November 7, 2016.

- **(b) Affected ADs**

  None.

- **(c) Applicability**

  This AD applies to The Boeing Company Model 727, 727–100, 727C, 727–100C, 727–200, and 727–200F series airplanes; certified in any category; equipped with Boeing body-mounted auxiliary fuel tanks.

- **(d) Subject**

  Air Transport Association (ATA) of America Code 28, Fuel.

- **(e) Unsafe Condition**

  This AD was prompted by fuel system reviews conducted by the manufacturer. We are issuing this AD to prevent ignition sources inside the body-mounted auxiliary fuel tanks, which, in combination with flammable fuel vapors, could result in a fuel tank explosion and consequent loss of the airplane.

- **(f) Compliance**

  Comply with this AD within the compliance times specified, unless already done.

- **(g) Modification**

  Within 12 months after the effective date of this AD, do the actions specified in either paragraph (g)(1) or (g)(2) of this AD, using a method approved in accordance with the procedures specified in paragraph (h) of this AD.

  1. Modify the fuel quantity indicating system (FQIS) to prevent development of an ignition source inside the body-mounted auxiliary fuel tanks due to electrical fault conditions.

  2. Deactivate the body-mounted auxiliary fuel tanks.

- **(h) Alternative Methods of Compliance (AMOCs)**

  1. The Manager, Seattle Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the ACO, send it to the attention of the person identified in paragraph (i) of this AD. Information may be emailed to: 9-ANM-Seattle-ACO-AMOC-Requests@faa.gov.

  2. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

  3. An AMOC that provides an acceptable level of safety may be used for any repair, modification, or alteration required by this AD if it is approved by the Boeing Commercial Airplanes Organization Designation Authorization (ODA) that has been authorized by the Manager, Seattle ACO, to make those findings. To be approved, the repair method, modification deviation, or alteration deviation must meet the certification basis of the airplane and the approval must specifically refer to this AD.

- **(i) Related Information**

  For more information about this AD, contact Jon Regimbala, Aerospace Engineer, Propulsion Branch, ANM–140S, FAA, Seattle ACO, 1601 Lind Avenue SW., Renton, WA 98057–5556; phone: 425–917–6506; fax: 425–917–8590; email: Jon.Regimbala@faa.gov.

Issued in Renton, Washington, on August 30, 2016.

Michael Kaszycki,
Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2016–21397 Filed 9–22–16; 8:45 am]

**DEPARTMENT OF TRANSPORTATION**

**Federal Aviation Administration**

14 CFR Part 39


**RIN 2120-AA64**

**Airworthiness Directives; B–N Group Ltd. Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** We propose to adopt a new airworthiness directive (AD) for B–N Group Ltd. Models BN–2, BN–2A, BN–2A–2–2, BN–2–2A–3, BN–2A–6, BN–2A–8, BN–2A–9, BN–2A–20, BN–2A–21, BN–2A–26, BN–2A–27, BN–2B–20, BN–2B–21, BN–2B–26, BN–2B–27, BN–2T–4R, BN–2T, BN2A MK. III, BN2A MK. III–2, and BN2A MK. III–3 (all models on Type Certificate Data Sheets A17EU and A29EU) airplanes that would supersede AD 2016–06–01. This proposed AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as cracks in the inner shell of certain pitot/static pressure heads. We are issuing this proposed AD to change the model applicability due to errors found in AD 2016–06–01.

**DATES:** We must receive comments on this proposed AD by November 7, 2016.

**ADDRESSES:** You may send comments by any of the following methods:

- Fax: (202) 493–2251.
- Hand Delivery: U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590,
between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. For service information identified in this proposed AD, contact Britten-Norman Aircraft Limited, Commodore House, Mountbatten Business Centre, Millbrook Road East, Southampton SO15 1HY, United Kingdom; telephone: +44 20 3371 4000; fax: +44 20 3371 4001; email: info@bnaircraft.com; Internet: http://www.britten-norman.com/customer-support/. You may review copies of the referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329–4148.

Exempted AD Docket

You may examine the AD docket on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA–2016–9160; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647–5527) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT:

Raymond Johnston, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust Road Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4159; fax: (816) 329–3047; email: raymond.johnston@faa.gov.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the ADDRESSES section. Include “Docket No. FAA–2016–9160; Directorate Identifier 2016–CE–022–AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD because of those comments.

We will post all comments we receive, without change, to http:// regulations.gov, including any personal information you provide. We will also post a listing identifying each substantive verbal contact we receive about this proposed AD.

Discussion


In 2005, occurrences were reported of finding cracks in the inner shell of certain pitot/static pressure heads, Part Number (P/N) DU130–24.

This condition, if not detected and corrected, could lead to incorrect readings on the pressure instrumentation, e.g. altimeters, vertical speed indicators (rate-of-climb) and airspeed indicators, possibly resulting in reduced control of the aeroplane.

To address this potential unsafe condition, B–N Group issued Service Bulletin (SB) 310 to provide inspection and test instructions. Consequently, CAA UK issued AD G–2005–0003 (EASA approval 2005–0447) to require repetitive inspections and leak tests and, depending on findings, accomplishment of applicable corrective action(s).

Subsequently, B–N Group published SB 310 issue 2, prompting EASA to issue AD 2006–0143 making reference to SB 310 at issue 2, while the publication of BNA SB 310 issue 3 prompted EASA AD 2006–0143R1, introducing BNA modification (mod) NB–M–1728 (new pitot/static pressure head not affected by the AD requirements) as optional corrective action(s).

Since that AD was issued, operators have reported a number of premature failures of the affected P/N DU130–24 pitot-static probes.

Prompted by these reports, BNA issued SB 310 issue 4 to reduce the interval for the inspections and leak tests.

Since we issued AD 2016–06–01, errors were discovered in the model applicability after issuance. This proposed AD adds Models BN–2T and BN–2T–4R, removes nonexistent Model BN2B, and removes duplicate listings of BN2A and BN2A MK.III.

Related Service Information Under 1 CFR Part 51

B–N Group Ltd. has issued Britten-Norman Service Bulletin Number SB 310, Issue 4, dated September 25, 2015. The service information describes procedures for leak tests and, if necessary, replacement of the pitot/static pressure head. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the ADDRESSES section of this NPRM.

FAA’s Determination and Requirements of the Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with this State of Design Authority, they have notified us of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all information and determined the unsafe condition exists and is likely to exist or develop on other products of the same type design.

Costs of Compliance

We estimate that this proposed AD will affect 93 products of U.S. registry. We also estimate that it would take about 1 work-hour per product to comply with the basic requirements of this proposed AD. The average labor rate is $85 per work-hour.

Based on these figures, we estimate the cost of this proposed AD on U.S. operators to be $7,905, or $85 per product.

In addition, we estimate that any necessary follow-on actions would take about 2 work-hours and require parts costing $10,000, for a cost of $10,170 per product. We have no way of determining the number of products that may need these actions.

The cost impact of this AD is the same as that presented in AD 2016–06–01.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.
Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation: (1) Is not a “significant regulatory action” under Executive Order 12866, (2) Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979), (3) Will not affect intrastate aviation in Alaska, and (4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by removing Amendment 39–18432 (81 FR 13717; March 15, 2016), and adding the following new AD:


(a) Comments Due Date

We must receive comments by November 7, 2016.

(b) Affected ADs

This AD replaces AD 2016–06–01, Amendment 39–18432 (81 FR 13717; March 15, 2016).

(c) Applicability


(d) Subject


(e) Reason

This AD was prompted by mandatory continuing airworthiness information (MCAI) issued by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as cracks in the inner shell of certain pitot/static pressure heads. We are issuing this proposed AD to change the model applicability due to errors found in AD 2016–06–01.

(f) Actions and Compliance

Unless already done, do the following actions in paragraphs (f)(1) through (5) of this AD:

(1) For all airplanes that are equipped with pitot/static pressure head part number (P/N) DU130–24, except Models BN–2T and BN–2T–4R: Within 50 hours time-in-service (TIS) after April 19, 2016 (the effective date retained from AD 2016–06–01) and repetitively thereafter at intervals not to exceed 50 hours TIS, inspect the pitot/static pressure head for cracks and/or separation and perform a leak test following the procedures in the section of Britten-Norman Service Bulletin SB 310, Issue 4, dated September 25, 2015.

(2) For Models BN–2T and BN–2T–4R that are equipped with pitot/static pressure head part number (P/N) DU130–24: Within 50 hours TIS after the effective date of this AD and repetitively thereafter at intervals not to exceed 50 hours TIS, inspect the pitot/static pressure head for cracks and/or separation and perform a leak test following the procedures in the section of Britten-Norman Service Bulletin SB 310, Issue 4, dated September 25, 2015.

(3) For all airplanes equipped with pitot/static pressure head part number (P/N) DU130–24: If, during an inspection or test required in paragraph (f)(1) or (2) of this AD discrepancies are found, before further flight, replace the pitot/static pressure head with an airworthy part.

(4) For all airplanes equipped with pitot/static pressure head part number (P/N) DU130–24: Corrections performed on airplanes as required in paragraph (f)(3) of this AD do not constitute terminating action for the repetitive actions required in paragraph (f)(1) or (2) of this AD.

(5) For all airplanes not equipped with a pitot/static pressure head P/N DU130–24 on the effective date of this AD: After April 19, 2016 (the effective date retained from AD 2016–06–01), do not install a pitot/static pressure head P/N DU130–24.

(g) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, Standards Office, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Raymond Johnston, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4159; fax: (816) 329–4090; email: Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) Airworthy Product: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(b) Related Information

Refer to MCAI European Aviation Safety Agency (EASA) AD No.: 2015–0184, dated September 1, 2015; for related information. You may examine the MCAI on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA–2016–9160. For service information related to this AD, contact Britten-Norman Aircraft Limited, Commodore House, Mountbatten Business Centre, Millbrook Road East, Southampton SO15 1HY, United Kingdom; telephone: +44 20 3371 4000; fax: +44 20 3371 4001; email: info@bnaircraft.com; Internet: http://www.britten-norman.com/customer-support/. You may review copies of the referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329–4148.

Issued in Kansas City, Missouri, on September 16, 2016.

Pat Mullen,
Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2016–22831 Filed 9–22–16; 8:45 am]