accordance with the procedures specified in paragraph (l)(2) of this AD.

(i) No Terminating Action for Inspections

Accomplishing corrective actions on an airplane as required by paragraph (g) or (h)(2) of this AD does not constitute terminating action for the repetitive actions required by paragraph (g) of this AD.

(j) Terminating Action

Accomplishment of the initial inspection required by paragraph (g) of this AD constitutes terminating action for the actions required by paragraph (h) of AD 2011–10–06.

(k) Credit for Previous Actions

This paragraph provides credit for actions required by paragraph (g) of this AD, if those actions were performed before the effective date of this AD using the service information specified in Airbus Service Bulletin A310–57–2050, Revision 03, dated December 19, 2014.

(l) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, International Section, Transport Standards Branch, FAA, has the authority to approve AMOCs for this AD. If requested using the procedures found in 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Section, send it to the attention of the person identified in paragraph (n)(2) of this AD. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(2) Contacting the Manufacturer: For any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, International Section, Transport Standards Branch, FAA; or the European Aviation Safety Agency (EASA); or Airbus’s EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(3) Required for Compliance (RC): Except as required by paragraph (h)(2) of this AD, if any service information contains procedures or tests that are identified as RC, those procedures and tests must be done to comply with this AD; any procedures or tests that are not identified as RC are recommended. Those procedures and tests that are not identified as RC may be deviated from using accepted methods in accordance with the operator’s maintenance or inspection program without obtaining approval of an AMOC. Provided the procedures and tests identified as RC can be done and the airplane can be put back in an airworthy condition. Any substitutions or changes to procedures or tests identified as RC require approval of an AMOC.

(m) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) EASA Airworthiness Directive 2016–0196, dated September 30, 2016, for related information. This MCAI may be found in the AD docket on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA–2017–0560.


(n) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(3) Service information identified in this AD is not incorporated by reference is available at the addresses specified in paragraphs (n)(3) and (n)(4) of this AD.

(3) Service information identified in this AD is not incorporated by reference is available at the addresses specified in paragraphs (n)(3) and (n)(4) of this AD.

(n) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(3) Service information identified in this AD is not incorporated by reference is available at the addresses specified in paragraphs (n)(3) and (n)(4) of this AD.

(4) You may view this service information at the FAA, Transport Standards Branch, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202–741–6030, or go to: http:// www.archives.gov/federal-register/cfr/ibr-locations.html.

Issued in Renton, Washington, on August 31, 2017.

Dione Palermo,
Acting Director, System Oversight Division, Aircraft Certification Service.

FR Doc. 2017–19042 Filed 9–13–17; 8:45 am
BILLING CODE 4910–13–P
Examiner the AD Docket

You may examine the AD docket on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA–2016–0451 or in the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (telephone 800–647–5527) is Docket Management Facility, U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590.


Supplementary Information:

Discussion


Airbus has issued Service Bulletin (SB) A300–53–6125, dated March 17, 2015, and Service Bulletin A300–53–6178, dated March 17, 2015. The service information describes procedures for the modification of certain upper frame foot fittings. These documents are distinct since they apply to airplanes in different configurations. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the ADDRESSES section.

Costs of Compliance

We estimate that this AD affects 65 airplanes of U.S. registry.

We estimate the following costs to comply with this AD:

The actions that were required by AD 2004–23–20 and retained in this AD take about 90 work-hours per product, at an average labor rate of $85 per work-hour. Required parts cost about $4,000 per product. Based on these figures, the estimated cost of the actions that were required by AD 2004–23–20 is $11,650 per product.

We also estimate that it will take up to 109 work-hours per product to comply with the new basic requirements of this AD. The average labor rate is $85 per work-hour. Required parts will cost up to $6,070 per product. Based on these figures, we estimate the cost of this AD on U.S. operators to be up to $996,775, or up to $15,335 per product.

We have received no definitive data that will enable us to provide cost estimates for the on-condition actions specified in this AD.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of
the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

This AD is issued in accordance with authority delegated by the Executive Director, Aircraft Certification Service, as authorized by FAA Order 8000.51C. In accordance with that order, issuance of ADs is normally a function of the Compliance and Airworthiness Division, but during this transition period, the Executive Director has delegated the authority to issue ADs applicable to transport category airplanes to the Director of the System Oversight Division.

Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

1. Is not a “significant regulatory action” under Executive Order 12866;
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska; and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

§ 39.13 [Amended]

(a) Effective Date

This AD is effective October 19, 2017.

(b) Affected ADs

This AD replaces AD 2004–23–20, Amendment 39–13875 (69 FR 68779, November 26, 2004), and adding the following new AD:


(c) Applicability


(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

TABLE 1 TO PARAGRAPH (g)(1) OF THIS AD—MODIFICATION SB A300–53–6125, REVISION 04

<table>
<thead>
<tr>
<th>Airplane usage</th>
<th>Initial compliance time</th>
</tr>
</thead>
<tbody>
<tr>
<td>AFT greater than 1.5</td>
<td>Within 10,200 flight cycles or 22,100 flight hours.</td>
</tr>
<tr>
<td>AFT equal to or less than 1.5</td>
<td>Within 11,000 flight cycles or 16,600 flight hours.</td>
</tr>
</tbody>
</table>

TABLE 2 TO PARAGRAPHS (g)(1) AND (g)(2) OF THIS AD—MODIFICATION SB A300–53–6178

<table>
<thead>
<tr>
<th>Airplane configuration</th>
<th>Initial compliance time</th>
</tr>
</thead>
<tbody>
<tr>
<td>Post-modification 12168</td>
<td>Within 27,100 flight cycles or 47,300 flight hours since the airplane’s first flight, whichever occurs first.</td>
</tr>
<tr>
<td>Post-SB A300–53–6125</td>
<td>Within 27,100 flight cycles or 47,300 flight hours after embodiment of SB A300–53–6125, whichever occurs first.</td>
</tr>
</tbody>
</table>

Airbus Service Bulletin A300–53–6178, dated March 17, 2015, specifies to contact Airbus for appropriate action, and specifies that action as “RC.” Before further flight,
accomplish corrective actions in accordance with the procedures specified in paragraph (l)(2) of this AD.

(b) Additional Post-Modification Actions

Prior to exceeding 24,100 total flight cycles or 42,000 total flight hours, whichever occurs first after doing the modification required by paragraph (g)(2) of this AD: Contact the Manager, International Section, Transport Standards Branch, FAA; or the European Aviation Safety Agency (EASA); or Airbus’s EASA Design Organization Approval (DOA); for instructions to do additional actions, and do those actions at the compliance times stated therein.

(i) Definition of AFT

For the purpose of this AD, to establish the applicable AFT for the actions required by paragraph (g)(1) of this AD, divide the total accumulated flight hours counted from take-off to touch-down by the total accumulated flight cycles as of the effective date of this AD.

(j) Credit for Previous Actions

This paragraph provides credit for the modification required by paragraph (g)(1) of this AD, if the modification was performed before the effective date of this AD using the service information specified in paragraph (j)(1), (j)(2), (j)(3), or (j)(4) of this AD:

(1) Airbus Service Bulletin A300–53–6125, dated November 8, 2000, which is not incorporated by reference in this AD.


(3) Airbus Service Bulletin A300–53–6125, Revision 02, dated February 25, 2005, which is not incorporated by reference in this AD.

(4) Airbus Service Bulletin A300–53–6125, Revision 03, dated September 13, 2011, which is not incorporated by reference in this AD.

(k) Exempt Airplanes

For airplanes on which Airbus Modification 12168 has been embodied in production: The modification required by paragraph (g)(1) of this AD is not required by this AD.

(l) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, International Section, Transport Standards Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Section, send it to the attention of the person identified in paragraph (m)(2) of this AD. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(2) Contacting the Manufacturer: As of the effective date of this AD, for any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, International Section, Transport Standards Branch, FAA; or EASA; or Airbus’s EASA DOA. If approved by the DOA, the approval must include the DOA-authorized signature.

(3) Required for Compliance (RC): Except as required by paragraphs (g)(1) and (g)(2) of this AD: If any service information contains procedures or tests that are identified as RC, those procedures and tests must be done to comply with this AD: Any procedures or tests that are not identified as RC are recommended. Those procedures and tests that are not identified as RC may be deviated from using accepted methods in accordance with the operator’s maintenance or inspection program without obtaining approval of an AMOC, provided the procedures and tests identified as RC can be done and the airplane can be put back in an airworthy condition. Any substitutions or changes to procedures or tests identified as RC require approval of an AMOC.

(m) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) EASA Airworthiness Directive 2016–0249, dated December 14, 2016; corrected January 10, 2017; for relevant information. This MCAI may be found in the AD docket on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA–2016–0451.


(3) Service information identified in this AD that is not incorporated by reference is available at the addresses specified in paragraphs (n)(3) and (n)(4) of this AD.

(n) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.


(3) For service information identified in this AD, contact Airbus SAS, Airworthiness Office—EAW, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; email account.airworth-eas@airbus.com; Internet http://www.airbus.com.

(4) You may view this service information at the FAA, Transport Standards Branch, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221. You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202–741–6030, or go to: http://www.archives.gov/federal-register/cfr/ibr-locations.html.

Issued in Renton, Washington, on August 29, 2017.

Jeffrey E. Duven,
Director, System Oversight Division, Aircraft Certification Service.

[FR Doc. 2017–19303 Filed 9–13–17; 8:45 am]
BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39


RIN 2120–AA64

Airworthiness Directives; Dassault Aviation Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for certain Dassault Aviation Model FALCON 900EX airplanes. This AD was prompted by a determination that new or more restrictive maintenance requirements and/or airworthiness limitations are necessary. This AD requires revising the maintenance or inspection program, as applicable, to incorporate new or more restrictive maintenance requirements and/or airworthiness limitations. We are issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective October 19, 2017.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of October 19, 2017.

ADDRESSES: For service information identified in this final rule, contact Dassault Falcon Jet Corporation, Teterboro Airport, P.O. Box 2000, South Hackensack, NJ 07606; telephone 201–440–6700; Internet http://www.dassaultfalcon.com. You may view this referenced service information at the FAA, Transport Standards Branch, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221. It is also available on the Internet at http://www.regulations.gov.