

(4) This AD does not adopt the provisions specified in paragraph (4) of EASA AD 2024–0208.

(5) This AD does not adopt the “Remarks” section of EASA AD 2024–0208.

(l) Retained Restrictions on Alternative Actions and Intervals From AD 2025–17–07, With a New Exception

This paragraph restates the requirements of paragraph (i) of AD 2025–17–07, with a new exception. Except as required by paragraph (m) of this AD, after the existing maintenance or inspection program has been revised as required by paragraph (j) of this AD, no alternative actions (*e.g.*, inspections) and intervals are allowed unless they are approved as specified in the provisions of the “Ref. Publications” section of EASA AD 2024–0208.

(m) New Revision of the Existing Maintenance or Inspection Program

Except as specified in paragraph (n) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, EASA AD 2025–0030, dated February 10, 2025 (EASA AD 2025–0030). Accomplishing the revision of the existing maintenance or inspection program required by this paragraph terminates the requirements of paragraphs (g) and (j) of this AD.

(n) Exceptions to EASA AD 2025–0030

(1) This AD does not adopt the requirements specified in paragraphs (1) and (2) of EASA AD 2025–0030.

(2) Paragraph (3) of EASA AD 2025–0030 specifies revising “the approved AMP” within 12 months after its effective date, but this AD requires revising the existing maintenance or inspection program, as applicable, within 90 days after the effective date of this AD.

(3) The initial compliance time for doing the tasks specified in paragraph (3) of EASA AD 2025–0030 is at the applicable “associated thresholds” as incorporated by the requirements of paragraph (3) of EASA AD 2025–0030, or within 90 days after the effective date of this AD, whichever occurs later.

(4) This AD does not adopt the provisions specified in paragraphs (4) and (5) of EASA AD 2025–0030.

(5) This AD does not adopt the “Remarks” section of EASA AD 2025–0030.

(6) This AD does not require incorporating Section 4, “Damage Tolerant—Airworthiness Limitations Items—tasks beyond MPPT,” of “the ALS” specified in EASA AD 2025–0030.

(o) New Provisions for Alternative Actions, Intervals, and CDCCLs

After the existing maintenance or inspection program has been revised as required by paragraph (m) of this AD, no alternative actions (*e.g.*, inspections), intervals, and CDCCLs are allowed unless they are approved as specified in the provisions of the “Ref. Publications” section of EASA AD 2025–0030.

(p) Additional AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, AIR–520, Continued Operational Safety Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or responsible Flight Standards Office, as appropriate. If sending information directly to the manager of the Continued Operational Safety Branch, send it to the attention of the person identified in paragraph (q) of this AD and email to: AMOC@faa.gov.

(i) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the responsible Flight Standards Office.

(ii) AMOCs approved previously for AD 2025–03–06 are approved as AMOCs for the corresponding provisions of EASA AD 2025–0030 that are required by paragraph (m) of this AD.

(2) *Contacting the Manufacturer*: For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, Continued Operational Safety Branch, FAA; or EASA; or Airbus SAS’s EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(q) Additional Information

For more information about this AD, contact Camille Seay, Aviation Safety Engineer, FAA, 2200 South 216th St., Des Moines, WA 98198; phone: 817–222–5149; email: Camille.L.Seay@faa.gov.

(r) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the material listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this material as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(3) The following material was approved for IBR on June 12, 2026.

(i) European Union Aviation Safety Agency (EASA) AD 2025–0030, dated February 10, 2025.

(ii) [Reserved]

(4) The following material was approved for IBR on October 1, 2025 (90 FR 41771, August 27, 2025).

(i) EASA AD 2024–0208, dated October 25, 2024.

(ii) [Reserved]

(5) The following material was approved for IBR on March 21, 2025 (90 FR 9595, February 14, 2025).

(i) EASA AD 2024–0031, dated January 31, 2024; corrected February 1, 2024.

(ii) [Reserved]

(6) For EASA material identified in this AD, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email ADs@easa.europa.eu. You may find this material on the EASA website at ad.easa.europa.eu.

(7) You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

(8) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit www.archives.gov/federal-register/cfr/ibr-locations or email fr.inspection@nara.gov.

Issued on April 23, 2026.

Lona C. Saccomando,

Acting Deputy Director, Integrated Certificate Management Division, Aircraft Certification Service.

[FR Doc. 2026–09169 Filed 5–7–26; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2025–5404; Project Identifier MCAI–2025–00424–T; Amendment 39–23325; AD 2026–09–04]

RIN 2120–AA64

Airworthiness Directives; Gulfstream Aerospace LP Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain Gulfstream Aerospace LP (GALP) Model Gulfstream G280 airplanes. This AD was prompted by reports of the accumulation of water in electrical connectors located in the aft fuselage directly below the empennage, resulting in empennage flight control related crew alerting system (CAS) messages. This AD requires retrofitting the flight controls empennage electrical harness. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective June 12, 2026.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of June 12, 2026.

ADDRESSES:

AD Docket: You may examine the AD docket at regulations.gov under Docket No. FAA–2025–5404; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the mandatory continuing airworthiness information (MCAI), any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

Material Incorporated by Reference:

- For Civil Aviation Authority of Israel (CAAI) material identified in this AD, contact CAAI, P.O. Box 1101, Golan Street, Airport City, 70100, Israel; telephone 972-3-9774665; fax 972-3-9774592; email aip@mot.gov.il. You may find this material on the CAAI website at www.gov.il/en/pages/israeli-airworthiness-directives.

- You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195. It is also available at regulations.gov under Docket No. FAA-2025-5404.

FOR FURTHER INFORMATION CONTACT:

Richard Bolden, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 404-474-5592; email richard.bolden@faa.gov.

SUPPLEMENTARY INFORMATION:

Background

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to certain GALP Model Gulfstream G280 airplanes. The NPRM was published in the **Federal Register** on January 7, 2026 (91 FR 457). The NPRM was prompted by AD ISR I-27-

2025-03-06 R1, dated August 28, 2025 (CAAI AD ISR I-27-2025-03-06 R1) (also referred to as the MCAI), issued by CAAI, which is the aviation authority for Israel. The MCAI states that several reports of empennage flight control related CAS messages have been attributed to the accumulation of water in electrical connectors located in the aft fuselage directly below the empennage. The unsafe condition, if not addressed, could, in combination with various specific failures or scenarios, result in loss of controllability of the airplane.

In the NPRM, the FAA proposed to require retrofitting the flight controls empennage electrical harness. The FAA is issuing this AD to address the unsafe condition on these products.

You may examine the MCAI in the AD docket at regulations.gov under Docket No. FAA-2025-5404.

Discussion of Final Airworthiness Directive

Comments

The FAA received no comments on the NPRM or on the determination of the cost to the public.

Conclusion

These products have been approved by the civil aviation authority of another country and are approved for operation in the United States. Pursuant to the

FAA's bilateral agreement with this State of Design Authority, that authority has notified the FAA of the unsafe condition described in the MCAI referenced above. The FAA reviewed the relevant data, considered any comments received, and determined that air safety requires adopting this AD as proposed. Accordingly, the FAA is issuing this AD to address the unsafe condition on these products. Except for minor editorial changes, this AD is adopted as proposed in the NPRM. None of the changes will increase the economic burden on any operator.

Material Incorporated by Reference Under 1 CFR Part 51

CAAI AD ISR I-27-2025-03-06 R1 specifies procedures for retrofitting the flight controls empennage electrical harness by replacing the backshells of electrical connectors at the vertical tail compartment.

This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

Costs of Compliance

The FAA estimates that this AD affects 140 airplanes of U.S. registry. The FAA estimates the following costs to comply with this AD:

ESTIMATED COSTS FOR REQUIRED ACTIONS

Labor cost	Parts cost	Cost per product	Cost on U.S. operators
80 work-hours × \$85 per hour = \$6,800	\$3,200	\$10,000	\$1,400,000

The FAA has included all known costs in its cost estimate. According to the manufacturer, however, some or all of the costs of this AD may be covered under warranty, thereby reducing the cost impact on affected operators.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds

necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Will not affect intrastate aviation in Alaska, and
- (3) Will not have a significant economic impact, positive or negative,

on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

2026-09-04 Gulfstream Aerospace LP:
Amendment 39-23325; Docket No. FAA-2025-5404; Project Identifier MCAI-2025-00424-T.

(a) Effective Date

This airworthiness directive (AD) is effective June 12, 2026.

(b) Affected ADs

None.

(c) Applicability

This AD applies to Gulfstream Aerospace LP Model Gulfstream G280 airplanes, certificated in any category, as identified in Civil Aviation Authority of Israel (CAAI) AD ISR I-27-2025-03-06 R1, dated August 28, 2025 (CAAI AD ISR I-27-2025-03-06 R1).

(d) Subject

Air Transport Association (ATA) of America Code 27, Flight Controls.

(e) Unsafe Condition

This AD was prompted by reports of the accumulation of water in electrical connectors located in the aft fuselage directly below the empennage, resulting in empennage flight control related crew alerting system (CAS) messages. The FAA is issuing this AD to address the accumulation of water in electrical connectors located in the aft fuselage directly below the empennage. The unsafe condition, if not addressed, could, in combination with various specific failures or scenarios, result in loss of controllability of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Requirements

Except as specified in paragraphs (h) and (i) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, CAAI AD ISR I-27-2025-03-06 R1.

(h) Exceptions to CAAI AD ISR I-27-2025-03-06 R1

Where CAAI AD ISR I-27-2025-03-06 R1 refers to its effective date, this AD requires using the effective date of this AD.

(i) No Reporting Requirement

Although the material referenced in CAAI AD ISR I-27-2025-03-06 R1 specifies to submit certain information to the manufacturer, this AD does not include that requirement.

(j) Additional AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19.

In accordance with 14 CFR 39.19, send your request to your principal inspector or responsible Flight Standards Office, as appropriate. If sending information directly to the manager of the International Validation Branch, send it to the attention of the person identified in paragraph (k) of this AD and email to: AMOC@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the responsible Flight Standards Office.

(2) *Contacting the Manufacturer*: For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, International Validation Branch, FAA; or CAAI; or CAAI's authorized Designee. If approved by the CAAI Designee, the approval must include the Designee's authorized signature.

(k) Additional Information

For more information about this AD, contact Richard Bolden, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; phone: 404-474-5592; email: richard.bolden@faa.gov.

(l) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference of the material listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this material as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(i) Civil Aviation Authority of Israel (CAAI) AD ISR I-27-2025-03-06 R1, dated August 28, 2025.

(ii) [Reserved]

(3) For CAAI material identified in this AD, contact CAAI, P.O. Box 1101, Golan Street, Airport City, 70100, Israel; telephone 972-3-9774665; fax 972-3-9774592; email aip@mot.gov.il. You may find this material on the CAAI website at www.gov.il/en/pages/israeli-airworthiness-directives.

(4) You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195.

(5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit www.archives.gov/federal-register/cfr/ibr-locations or email fr.inspection@nara.gov.

Issued on April 23, 2026.

Steven W. Thompson,

Acting Deputy Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2026-09170 Filed 5-7-26; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA-2026-3871; Project Identifier MCAI-2026-00247-T; Amendment 39-23327; AD 2026-09-06]

RIN 2120-AA64

Airworthiness Directives; Airbus SAS Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule; request for comments.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain Airbus SAS Model A319-153N airplanes; Model A320-251N, -252N, and -271N airplanes; and Model A321-251NX, -252NX, -271NX and -272NX airplanes. This AD was prompted by an Airbus supplier identifying a quality issue in production, which could result in potential deviations from the specified thickness of various fuselage panels. This AD requires, for certain airplanes, doing a "local thickness mapping," doing applicable additional instructions, a general visual inspection (GVI) of certain forward fuselage panels, a full panel thickness measurement, and applicable corrective actions. This AD also requires, for all airplanes, a dispatch restriction from using certain master minimum equipment list (M MEL) items, a prohibition to use certain structural repair manual (SRM) tasks. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective May 26, 2026.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of May 26, 2026.

The FAA must receive comments on this AD by June 22, 2026.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal*: Go to regulations.gov. Follow the instructions for submitting comments.

- *Fax*: 202-493-2251.

- *Mail*: U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

- *Hand Delivery*: Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.