

2026–20–07 Leonardo S.p.a.: Amendment 39–23480; Docket No. FAA–2026–11789; Project Identifier MCAI–2026–00908–R.

(a) Effective Date

This airworthiness directive (AD) is effective October 16, 2026.

(b) Affected ADs

None.

(c) Applicability

This AD applies to Leonardo S.p.a. Model AB139 and AW139 helicopters, certificated in any category, as identified in European Union Aviation Safety Agency AD 2026–0165, dated August 18, 2026 (EASA AD 2026–0165).

(d) Subject

Joint Aircraft System Component (JASC) Code 7800, Engine exhaust system.

(e) Unsafe Condition

This AD was prompted by a report of a nozzle exhaust duct clamp separating from the engine nozzle duct and coming to rest on the tail rotor (TR) drive shaft, which caused damage to the TR drive shaft. The FAA is issuing this AD to detect and address damage to the TR drive shaft, which could result in loss of control of the helicopter around the yaw axis.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Requirements

Except as specified in paragraphs (h) and (i) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, EASA AD 2026–0165.

(h) Exceptions to EASA AD 2026–0165

(1) Where EASA AD 2026–0165 refers to its effective date, this AD requires using the effective date of this AD.

(2) Where paragraph (1) of EASA AD 2026–0165 requires compliance in terms of flight hours, this AD requires using hours time-in-service.

(3) Where the material referenced in EASA AD 2026–0165 specifies “perform the detailed visual inspection for damage and crack(s)”, this AD requires replacing that text with “perform the detailed visual inspection for physical damage including, but not limited to, cracks, gouges, tears, bulges, fractures, distortions, and creases.”

(4) This AD does not adopt the “Remarks” section of EASA AD 2026–0165.

(i) No Reporting or Returning of Parts Requirement

Although EASA AD 2026–0165 and the material referenced in EASA AD 2026–0165 specifies to submit certain information and return parts to the manufacturer, this AD does not require those actions.

(j) Alternative Methods of Compliance (AMOCs)

(1) The Manager, International Validation Branch, FAA, has the authority to approve

AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the International Validation Branch, send it to the attention of the person identified in paragraph (k) of this AD and email to AMOC@faa.gov.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(k) Additional Information

For more information about this AD, contact Evan Weaver, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; phone: (316) 944–8910; email: evan.p.weaver@faa.gov.

(l) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference of the material listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this material as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) European Union Aviation Safety Agency (EASA) AD 2026–0165, dated August 18, 2026.

(ii) [Reserved]

(3) For EASA material identified in this AD, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; phone: +49 221 8999 000; email: ADs@easa.europa.eu; website: easa.europa.eu. You may find the EASA material on the EASA website at ad.easa.europa.eu.

(4) You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 10101 Hillwood Parkway, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call (817) 222–5110.

(5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit www.archives.gov/federal-register/cfr/ibr-locations or email fr.inspection@nara.gov.

Issued on September 24, 2026.

Steven W. Thompson,

Acting Deputy Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2026–20100 Filed 9–30–26; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2026–1328; Project Identifier MCAI–2025–00462–T; Amendment 39–23478; AD 2026–20–05]

RIN 2120–AA64

Airworthiness Directives; Airbus SAS Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is superseding Airworthiness Directive (AD) 2021–03–08, which applied to certain Airbus SAS Model A350–941 and –1041 airplanes. AD 2021–03–08 required repetitive inspections for migration of the bushings of the horizontal tail plane (HTP) lateral load fittings (LLFs) on the left- and right-hand sides and terminating repair or modification of any affected bushing. Since the FAA issued AD 2021–03–08, new occurrences of bushing migration on HTP LLFs were reported, and a determination was made that certain repairs can no longer be considered terminating action to the repetitive inspections. This AD continues to require the actions in AD 2021–03–08, removes a certain terminating action, and expands the applicability. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective November 5, 2026.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of November 5, 2026.

ADDRESSES:

AD Docket: You may examine the AD docket at regulations.gov under Docket No. FAA–2026–1328; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the mandatory continuing airworthiness information (MCAI), any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

Material Incorporated by Reference:

- For European Union Aviation Safety Agency (EASA) material identified in this AD, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221

8999 000; email ADs@easa.europa.eu. You may find this material on the EASA website at ad.easa.europa.eu.

- You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195. It is also available at regulations.gov under Docket No. FAA-2026-1328.

FOR FURTHER INFORMATION CONTACT:

Andrew Younglove, Aviation Safety Engineer, FAA, 2200 South 216th St., Des Moines, WA 98198; phone: 206-231-3644; email: andrew.e.younglove@faa.gov.

SUPPLEMENTARY INFORMATION:

Background

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to supersede AD 2021-03-08, Amendment 39-21411 (86 FR 16038, March 26, 2021) (AD 2021-03-08). AD 2021-03-08 applied to certain Airbus SAS Model A350-941 and -1041 airplanes. AD 2021-03-08 required repetitive inspections for migration of the bushings of the HTP LLFs on the left- and right-hand sides and terminating repair or modification of any affected bushing. The FAA issued AD 2021-03-08 to address combined corrosion and fatigue damage of the primary structure, possibly resulting in failure of the HTP LLF and damage to adjacent structure, which could result in reduced controllability of the airplane.

The NPRM was published in the **Federal Register** on February 17, 2026 (91 FR 7183). The NPRM was prompted by EASA AD 2025-0073, dated April 3, 2025 (EASA AD 2025-0073) (also referred to as the MCAI), issued by EASA, which is the Technical Agent for the Member States of the European Union. The MCAI states that new occurrences of bushing migration on HTP LLF have been reported on airplanes with modification 110669, which were not in the applicability of EASA AD 2020-0139R1, dated July 3, 2020 (EASA AD 2020-0139R1). Additionally, it has been determined that the repair instructions provided by Airbus Service Bulletin A350-55-P013 at any revision can no longer be considered terminating action for the repetitive inspections.

In the NPRM, the FAA proposed to continue to require the actions in AD 2021-03-08, remove a certain terminating action, and expand the applicability, as specified in EASA AD 2025-0073. The FAA is issuing this AD to address the unsafe condition on these products.

You may examine the MCAI in the AD docket at regulations.gov under Docket No. FAA-2026-1328.

Discussion of Final Airworthiness Directive

Comments

The FAA received a comment from the Air Line Pilots Association, International (ALPA) who supported the NPRM without change.

The FAA received an additional comment from The Foundation for Aviation Safety. The following presents the comment received on the NPRM and the FAA's response to the comment.

Request for Clarification of Extended Compliance Time

The Foundation for Aviation Safety asked why the rulemaking process took six years. The commenter further stated, if safety is the top priority, conducting immediate inspections of the bushings should have been the first step, rather than allowing operators to defer inspections for up to six years or thousands of flight cycles.

The FAA disagrees that the rulemaking process took six years. AD 2021-03-08 was published on March 26, 2021, to address the unsafe condition on all airplanes known to be affected by the unsafe condition at that time. After issuance of AD 2021-03-08, new occurrences of bushing migration were reported on airplanes that were excluded from the applicability of AD 2021-03-08, and it was also determined that certain repairs can no longer be considered terminating action to the repetitive inspections, as specified in EASA AD 2025-0073. For these reasons, the FAA initiated a new rulemaking to supersede AD 2021-03-08.

The FAA also disagrees that the initial inspection should be accomplished immediately. EASA is the State of Design Authority for Airbus SAS Model A350-941 and -1041 airplanes. Pursuant to the FAA's bilateral agreement with EASA, EASA notified the FAA of the unsafe condition described in EASA AD 2025-0073.

EASA, as the state of design authority for these airplanes, analyzed the data, considered the recommendations of the manufacturer, and determined that the initial inspection timeframe was appropriate based on the safety implications of the identified unsafe condition. The FAA concurs with EASA's determination. The FAA has not changed this AD in response to this comment.

Conclusion

These products have been approved by the civil aviation authority of another country and are approved for operation in the United States. Pursuant to the FAA's bilateral agreement with this State of Design Authority, that authority has notified the FAA of the unsafe condition described in the MCAI referenced above. The FAA reviewed the relevant data, considered any comments received, and determined that air safety requires adopting this AD as proposed. Accordingly, the FAA is issuing this AD to address the unsafe condition on these products. Except for minor editorial changes, this AD is adopted as proposed in the NPRM. None of the changes will increase the economic burden on any operator.

Material Incorporated by Reference Under 1 CFR Part 51

EASA AD 2025-0073 specifies procedures for repetitive detailed inspections for discrepancies (e.g., broken sealant and migration) of the bushings of the HTP LLF on the left- and right-hand sides, and repair or modification of any affected bushings. EASA AD 2025-0073 specifies that modification of affected bushings terminates the repetitive inspections. EASA AD 2025-0073 also requires reporting inspection results to Airbus. EASA AD 2025-0073 also provides, for certain airplanes, an optional method of compliance for doing the initial inspections. This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

Costs of Compliance

The FAA estimates that this AD affects 36 airplanes of U.S. registry. The FAA estimates the following costs to comply with this AD:

ESTIMATED COSTS FOR REQUIRED ACTIONS

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Retained actions from AD 2021–03–08.	10 work-hours × \$85 per hour = \$850.	\$0	\$850	\$30,600 per inspection cycle.

ESTIMATED COSTS FOR OPTIONAL ACTIONS

Labor cost	Parts cost	Cost per product
56 work-hours × \$85 per hour = \$4,760	\$23,000	\$27,760

The FAA estimates the following costs to do any necessary on-condition action that would be required based on

the results of any required actions. The FAA has no way of determining the

number of aircraft that might need this on-condition action:

ESTIMATED COSTS OF ON-CONDITION ACTIONS

Labor cost	Parts cost	Cost per product
Up to 93 work-hours × \$85 per hour = \$7,905	Up to \$4,480 (for four bushings)	Up to \$12,385.

Paperwork Reduction Act

A federal agency may not conduct or sponsor, and a person is not required to respond to, nor shall a person be subject to a penalty for failure to comply with a collection of information subject to the requirements of the Paperwork Reduction Act unless that collection of information displays a currently valid OMB Control Number. The OMB Control Number for this information collection is 2120–0056. Public reporting for this collection of information is estimated to take approximately 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. All responses to this collection of information are mandatory. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to: Information Collection Clearance Officer, Federal Aviation Administration, 10101 Hillwood Parkway, Fort Worth, TX 76177–1524.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency’s authority.

The FAA is issuing this rulemaking under the authority described in

Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a “significant regulatory action” under Executive Order 12866,
- (2) Will not affect intrastate aviation in Alaska, and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:
- Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by:
- a. Removing Airworthiness Directive (AD) 2021–03–08, Amendment 39–21411 (86 FR 16038, March 26, 2021); and
- b. Adding the following new AD:
- 2026–20–05 Airbus SAS: Amendment 39–23478; Docket No. FAA–2026–1328; Project Identifier MCAI–2025–00462–T.

(a) Effective Date

This airworthiness directive (AD) is effective November 5, 2026.

(b) Affected ADs

This AD replaces AD 2021–03–08, Amendment 39–21411 (86 FR 16038, March 26, 2021) (AD 2021–03–08).

(c) Applicability

This AD applies to Airbus SAS Model A350–941 and –1041 airplanes, certificated in any category, as identified in European Union Aviation Safety Agency (EASA) AD 2025–0073, dated April 3, 2025 (EASA AD 2025–0073).

(d) Subject

Air Transport Association (ATA) of America Code 55, Stabilizers.

(e) Unsafe Condition

This AD was prompted by reports of migration of the bushings of the horizontal tail plane (HTP) lateral load fittings (LLFs) on the left- and right-hand sides during flight test. The FAA is issuing this AD to address combined corrosion and fatigue damage of the primary structure, possibly resulting in failure of the HTP LLF and damage to adjacent structure. The unsafe condition, if not addressed, could result in reduced controllability of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Requirements

Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, EASA AD 2025–0073.

(h) Exceptions to EASA AD 2025–0073

(1) Where EASA AD 2025–0073 refers to its effective date, this AD requires using the effective date of this AD.

(2) Where paragraphs (3) and (6) of EASA AD 2025–0073 refer to discrepancies, this AD defines discrepancies as broken sealant and bush migration.

(3) This AD does not adopt the “Remarks” section of EASA AD 2025–0073.

(i) Additional AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, AIR–520, Continued Operational Safety Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or responsible Flight Standards Office, as appropriate. If sending information directly to the manager of the Continued Operational Safety Branch, send it to the attention of the person identified in paragraph (j) of this AD and email to: AMOC@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the responsible Flight Standards Office.

(2) *Contacting the Manufacturer*: For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, AIR–520, Continued Operational Safety Branch, FAA; or EASA; or Airbus SAS’s EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(3) *Required for Compliance (RC)*: Except as required by paragraph (i)(2) of this AD, if any material contains procedures or tests that are identified as RC, those procedures and tests must be done to comply with this AD; any procedures or tests that are not identified as RC are recommended. Those procedures and tests that are not identified as RC may be deviated from using accepted methods in accordance with the operator’s maintenance or inspection program without obtaining

approval of an AMOC, provided the procedures and tests identified as RC can be done and the airplane can be put back in an airworthy condition. Any substitutions or changes to procedures or tests identified as RC require approval of an AMOC.

(j) Additional Information

For more information about this AD, contact Andrew Younglove, Aviation Safety Engineer, FAA, 2200 South 216th St., Des Moines, WA 98198; phone: 206–231–3644; email: andrew.e.younglove@faa.gov.

(k) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference of the material listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this material as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(i) European Union Aviation Safety Agency (EASA) AD 2025–0073, dated April 3, 2025.

(ii) [Reserved]

(3) For EASA material identified in this AD, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email ADS@easa.europa.eu. You may find this material on the EASA website at ad.easa.europa.eu.

(4) You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

(5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit www.archives.gov/federal-register/cfr/ibr-locations or email fr.inspection@nara.gov.

Issued on September 23, 2026.

Brian Knaup,

Acting Deputy Director, Integrated Certificate Management Division, Aircraft Certification Service.

[FR Doc. 2026–20119 Filed 9–30–26; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA–2026–4657; Project Identifier MCAI–2024–00383–R; Amendment 39–23481; AD 2026–20–08]

RIN 2120–AA64

Airworthiness Directives; Airbus Helicopters

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for all Airbus Helicopters Model AS355E, AS355F, AS355F1, AS355F2, and

AS355N helicopters. This AD was prompted by the determination that the tail rotor drive fan wheels (fan wheels) and the impeller second stage have been incorrectly identified during production. This AD requires verifying the serial number and amendment of the log cards of the affected parts, and depending on the findings, performing a one-time borescope inspection to verify consistency between the log cards and the affected parts. Depending on the results of the inspection, this AD requires replacing or re-identifying certain parts. This AD also prohibits the installation of certain fan wheels and impeller second stages on any helicopter. The FAA is issuing this AD to address the unsafe conditions on these products.

DATES: This AD is effective November 5, 2026.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of November 5, 2026.

ADDRESSES:

AD Docket: You may examine the AD docket at regulations.gov under Docket No. FAA–2026–4657; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the mandatory continuing airworthiness information (MCAI), any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

Material Incorporated by Reference:

- For European Union Aviation Safety Agency (EASA) material identified in this AD, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; phone: +49 221 8999 000; email: ADS@easa.europa.eu; website: easa.europa.eu. You may find the EASA material on the EASA website at ad.easa.europa.eu.

- You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 10101 Hillwood Parkway, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call (817) 222–5110. It is also available at regulations.gov under Docket No. FAA–2026–4657.

FOR FURTHER INFORMATION CONTACT:

Andrew Gronau, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; phone: (847) 294–7378; email: andrew.s.gronau@faa.gov.

SUPPLEMENTARY INFORMATION: